

Date: Monday, 3/13/2006 7:31:18 AM
 User: Kim Johnston

Process Sheet

Customer	: CU-DAR001 Dart Helicopters Services	Drawing Name	: SLIDE BAR
Job Number	: 26219		
Estimate Number	: 11085		
P.O. Number	: N/A	Part Number	: D30111
This Issue	: 3/13/2006 S.O. No. : N/A	Drawing Number	: D3011 REV A
Prsht Rev.	: NC	Project Number	: N/A
First Issue	: N/A	Drawing Revision	: A
Previous Run	: N/A	Material	: N/A
Written By	: See COMMENT BELOW	Due Date	: 3/30/2006 Qty: 5 Um: Each
Checked & Approved By	: JF 06.03.13		
Comment	: Est. C 02.06.09 Added D6202 at step 2 NG		

Additional Product

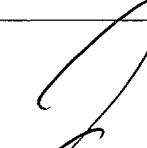
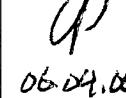
Job Number:



Seq. #:	Machine Or Operation:	Description :	
1.0	D6202	I Beam Extrusion	
		Comment: Qty.: 1.0000 f(s)/Unit Total : 5.0000 f(s) I Beam Extrusion Material: 6061-T6 (QQ-A-200/8) 'I' Beam Extrusion 4" x 2.796" x 0.326" Batch # 14742	J-F 06/04/05
2.0	BAND SAW	BAND SAW	
		Comment: BAND SAW Cut Blanks: 26.57"	JF 06/04/05
3.0	HAAS1	HAAS CNC VERTICAL MACHINING #1	
		Comment: HAAS CNC VERTICAL MACHINING #1 Ensure Batch Number programmed matches this W/O Machine as per folio FA129	JF / JL 06.04.07
4.0	QC2	INSPECT PARTS AS THEY COME OFF MACHINE	
		Comment: INSPECT PARTS AS THEY COME OFF MACHINE	/ JL 06.04.07
5.0	SMALL FAB 1	SMALL & MEDIUM FAB RESOURCE 1	
		Comment: SMALL & MEDIUM FAB RESOURCE 1 Deburr	JL 06.04.07

W/O:		WORK ORDER CHANGES					
DATE	STEP	PROCEDURE CHANGE	By	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector

Part No: _____ PAR #: _____ Fault Category: _____ NCR: Yes No DQA:  Date: 06/04/19
 QA: N/C Closed: _____ Date: _____

NCR:		WORK ORDER NON-CONFORMANCE (NCR)						
DATE	STEP	Description of NC Section A	Corrective Action Section B			Verification Section C	Approval Chief Eng	Approval QC Inspector
			Initial Chief Eng	Action Description Chief Eng	Sign & Date			
06/04/05	3	Not the good batch# Batch wasn't changed in program and ran previous B# (B17087)	 <u>QSI-12</u>	Check the batch number in the program before machining the piece. Type new B# Scrap, destroy & replace.	J-F 06/04/05 06.04.05		 QSI-12	 06.04.05
06.04.06	3	Dimension 2.125 is 2.097 Dim 2.500 is 2.465 see attached email and diagrams	 06.04.06 per QSI-042	PART IS OK PER ATTACHED ANALYSIS AND EMAIL	J-L 06.04.06 06.04.11		 06.04.06 per QSI-042	 06.04.11

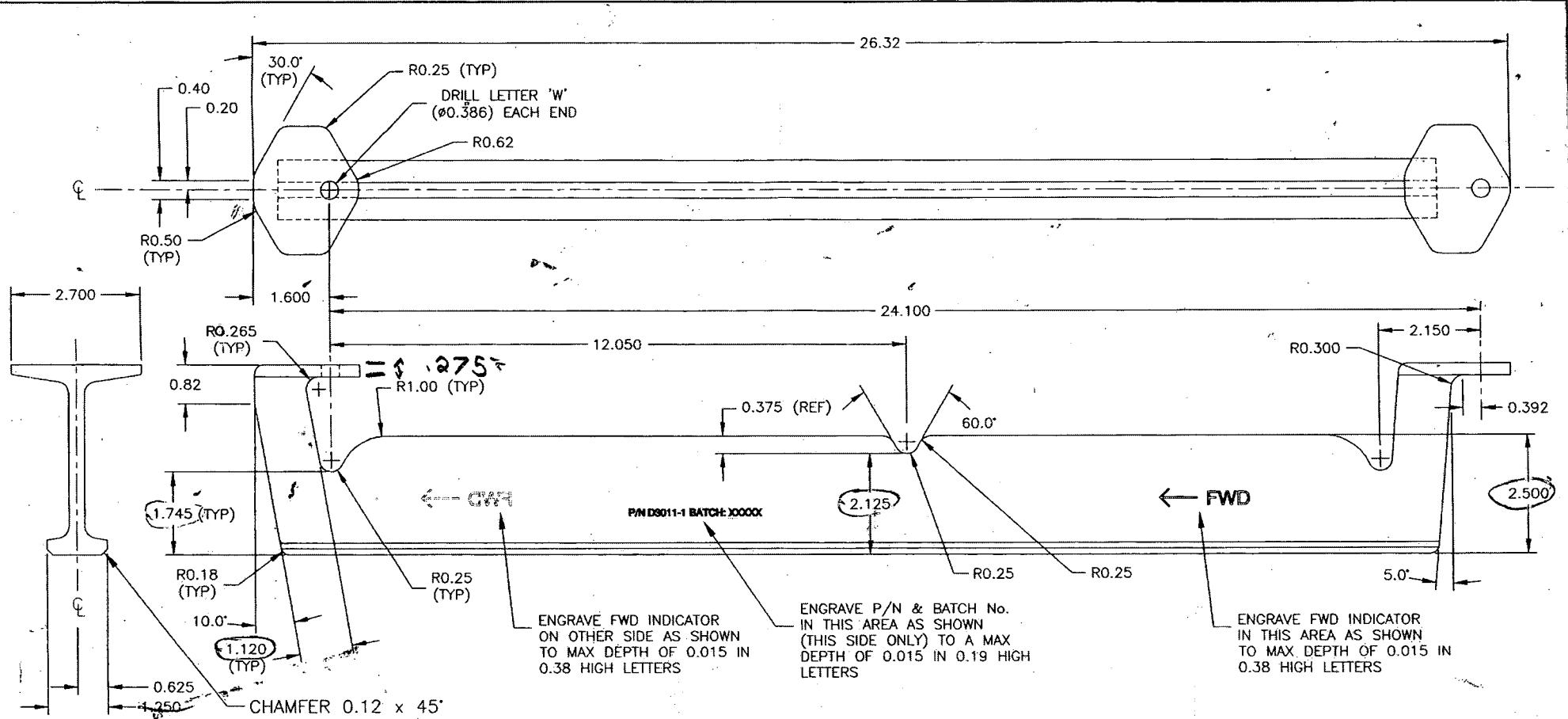
NOTE: Date & initial all entries

W/O:		WORK ORDER CHANGES					
DATE	STEP	PROCEDURE CHANGE	By	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector

Part No: _____ PAR #: _____ Fault Category: _____ NCR: Yes No DQA: _____ Date: _____
 QA: N/C Closed: _____ Date: _____

NCR:		WORK ORDER NON-CONFORMANCE (NCR)						
DATE	STEP	Description of NC Section A	Corrective Action Section B			Verification Section C	Approval Chief Eng	Approval QC Inspector
			Initial Chief Eng	Action Description Chief Eng	Sign & Date			

NOTE: Date & initial all entries



RELEASED

DA COPY
01.04.09

D3011-1

MANUFACTURE FROM D6202-027 EXTRUSION

BREAK ALL SHARP EDGES 0.010-0.020

FINISH: ACID ETCH AND ALODINE PER DART QSI 005 4.1

POWDER COAT WHITE (4.3.5.1) PER DART QSI 005 4.3

TOLERANCES ARE PER DART QSI 018 UNLESS OTHERWISE NOTED

ALL DIMENSIONS ARE IN INCHES

NO 2629
WORK ORDER
TO AMENDMENT
ROUTED COPY
TO ENGINEERING
JRN TO
HOLDING
SHOP COPY
TO OUT NOTICE

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A	01.03.29	NEW ISSUE
DESIGN	DRAWN BY <i>CP</i>	DART AEROSPACE LTD. HAWKSBURY, ONTARIO, CANADA
CHECKED	APPROVED	DRAWING NO. D3011
DATE	01.03.29	REV. A SHEET 1 OF 1 SCALE 1:2

RAPPAL SLIDE BAR

Chris Provencal

From: David Shepherd [davids@dartaero.com]
Sent: April 6, 2006 2:15 PM
To: Chris Provencal
Subject: Re: ncr D3011-1

Chris,

I have reviewed the analysis you sent by fax and I agree that the parts mentioned below are an acceptable deviation. Please attach a copy of the analysis and this email to the work order.

Thanks,
David

----- Original Message -----

From: "Chris Provencal" <cprovencal@dartaero.com>
To: <davids@dartaero.com>
Sent: Thursday, April 06, 2006 11:55 AM
Subject: ncr D3011-1

>
> Dave,
>
> The D3011-1 Rappel Slide Bar. The dimension 2.125" on the dwg is 2.097,
and
> the height 2.500" is 2.465" at its minimum. I did a calculation based on
> SUB2-D205-523 Rev. A page 2, and the margin is still positive. I will fax
> you that calculation, which I did on the photocopied page 2 of the
> aforementioned document.
>
> Is this acceptable?
>
> Sincerely,
> Chris Provencal
> DART Aerospace Ltd.
> Email...cprovencal@dartaero.com
> Phone...613-632-3336
> Fax.....613-632-4443
>
>

DESIGN	DRAWN BY	DART AEROSPACE LTD HAWKESBURY, ONTARIO, CANADA	
CHECKED	APPROVED	DRAWING NO.	REV. A SUB2-D205-523 (REF) SHEET 2 OF 3
DATE	01.03.30	TITLE	SCALE NTS SUBSTANTIATION REPORT

FROM PAGE 18 OF SR205-523, THE ROOF HARPOON @ STA 105.1 IS RATED FOR 1500 LB ULTIMATE LOAD.

$$\therefore P = (317 \text{ lb})(2.5)(1.5) = 1189 \text{ lb}$$

$$MS = \frac{1500 \text{ lb}}{1189 \text{ lb}} - 1 = \underline{0.26} \leftarrow \text{OK}$$

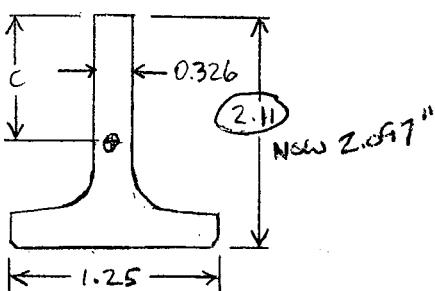
4.0 D2224 ANCHOR ANALYSIS

PER SUB1-D205-523 REV. A, THE D2224 ANCHOR WAS SUBSTANTIATED FOR A FACTORED LOAD OF 1294 LB FOR THE D205-523-011 INSTALLATION. IN THE CASE OF THE D205-523-013 INSTALLATION, THE ANCHOR MUST BE SUBSTANTIATED FOR $F = (600 \text{ lb})(2.5)(1.5) = 2250 \text{ lb}$. IF $F = 2250 \text{ lb}$, THE MARGINS IN SUB1-D205-523 GET RE-CALCULATED AS FOLLOWS:

$$\left. \begin{array}{l} MS1 = 0.11 \\ MS2 = 20.4 \\ MS3 = 3.7 \\ MS4 = 3.5 \end{array} \right\} \text{ALL POSITIVE, ALL OK}$$

5.0 D3011-1 SLIDE BAR ANALYSIS

IN COMPARISON TO THE D1005 SLIDE BAR, THE D3011-1 SLIDE BAR HAS AN INCREASED SECTION TO HANDLE A 500 LB WORKING LOAD AT POINT B.



$$F_{tu} = 38 \text{ ksi} \quad (\text{QQ-A-200 / 8})$$

$$\left. \begin{array}{l} C = 1.33'' \\ I = 0.42 \text{ in}^4 \end{array} \right\} \text{FROM AUTOCAD} \quad \left. \begin{array}{l} C = 1.33'' \\ I = 0.416 \text{ in}^4 \end{array} \right\} \begin{array}{l} (\text{Autocad}) \\ (\text{Autocad}) \end{array}$$

$$P = (500 \text{ lb})(2.5)(1.5) = \underline{1875 \text{ lb}} \quad P = 1875 \text{ lb}$$

$$M = \frac{PL}{4} = \frac{(1875 \text{ lb})(24.10'')}{4} = \underline{11297 \text{ lb-in}} \quad M = 11297 \text{ lb-in}$$

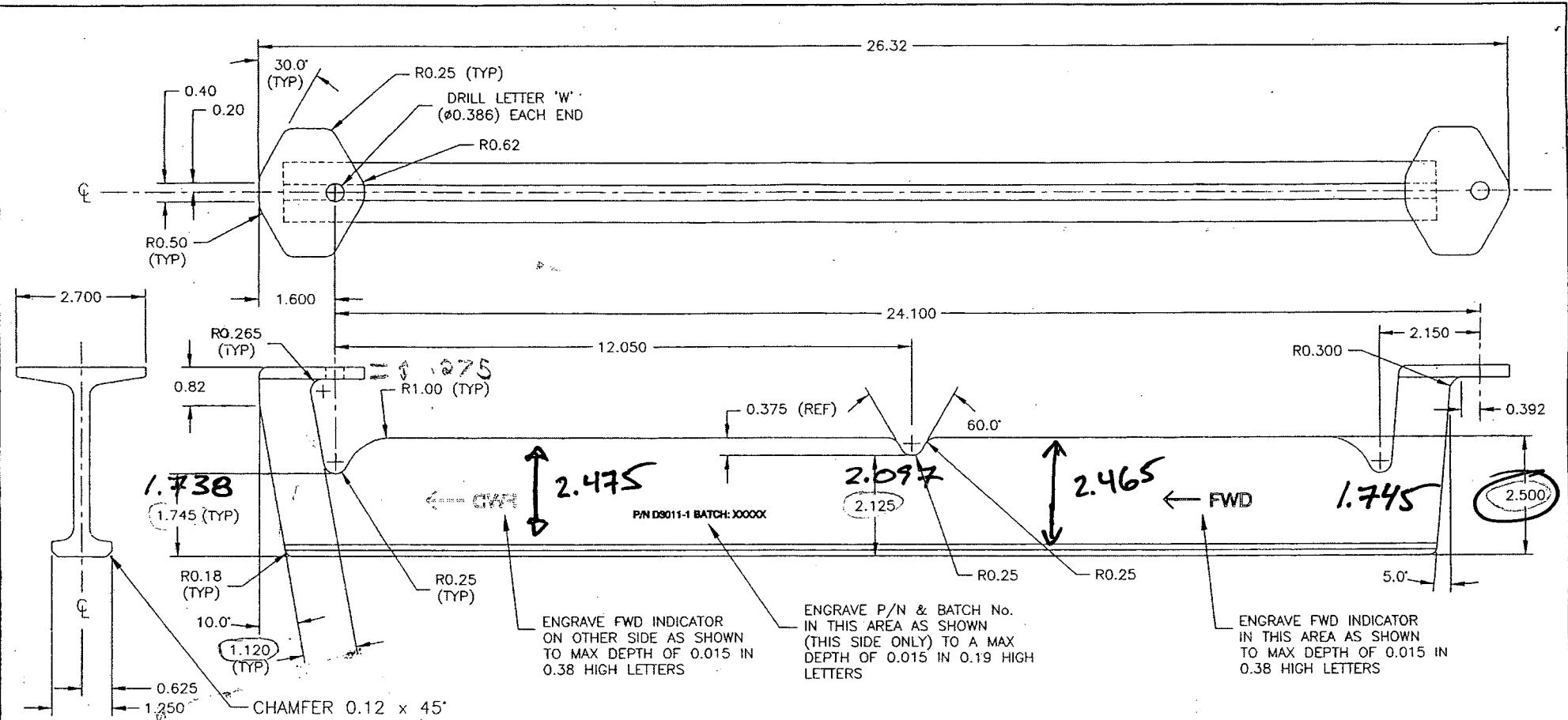
$$\sigma_{cr} = \frac{Mc}{I} = \frac{(11297 \text{ lb-in})(1.33'')}{0.42 \text{ in}^4} = 35.8 \text{ ksi} \quad T = \frac{11297 \cdot 1.33}{0.416}$$

$$MS = \frac{F_{tu}}{\sigma_{cr}} - 1 = \frac{38 \text{ ksi}}{35.8 \text{ ksi}} - 1 = \underline{0.06} \leftarrow \text{OK} \quad = 36118$$

$$MS = \frac{38}{36118} - 1 = 0.05$$

(P) 06.04.06

OK



REFERENCE ONLY

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01.04.09

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D3011-1
MANUFACTURE FROM D6202-027 EXTRUSION

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A	01.03.29	NEW ISSUE
DESIGN #	DRAWN BY CP	DART DART AEROSPACE LTD. HAWKSBURY, ONTARIO, CANADA
CHECKED #	APPROVED #	DRAWING NO. D3011 REV. A SHEET 1 OF 1
DATE 01.03.29	TITLE RAPPEL SLIDE BAR	SCALE 1:2

NO. 26279
WORK ORDER
JB ENGINEERING
ALL DIMENSIONS ARE IN INCHES
NOTICE TO AMENDMENT
NO. 100 ROLLED COPY
JRN TO
TOP COPY

DART AEROSPACE LTD

Work Order: 26219

Description: Slide bar

Part Number: D3011-

Inspection Dwg: D3011 Rev: A

Page 1 of 1

FIRST ARTICLE INSPECTION CHECKLIST

First Article Prototype

Measured by: T.J.

Audited by: M Q

Prototype Approval:

Date: 08/04/06

Date: 8/1/13

Date:

Rev	Date	Change
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BRUNNEN